**To:** Bay Area Compliance Lab Corporation

**From:** Jason Funderburk, LiveTV

**Date:** 3/27/2003

**Re:** WADL Aircraft Segment Installation Description

I lieu of an Installation or Users Manual this document will describe the installation of the Aircraft Segment of the Wireless Aircraft Data Link (WADL) system manufactured and installed on commercial aircraft by LiveTV. The WADL Aircraft segment is to be tested by BACL for compliance under FCC Part 15C.

The WADL Aircraft Segment is certified by the FAA for installation on Airbus A-320 and A-319 aircraft. Currently held certifications are:

**A-320** Supplemental Type Certification ST01447NY Granted: 11 February 2002

A-319 Supplemental Type Certification ST00788SE Last Amended: 3 December 2002

During the installation process all physical installation steps, inspections, and checkout procedures are documented on LiveTV generated Work Orders. An example set of work orders for the A-319 are shown in an attachment to this document. The installation and certification processes are described in general in the sections below.

#### Statement of Professional Installation and Certification

<u>Aircraft Segment Statement</u>: The WADL Aircraft Segment must be professionally installed because each installation is custom suited to meet the demands of the specific environment. The FAA requires a Supplemental Type Certificate (STC) for any changes to an existing aircraft type. When an airline acquires WADL Aircraft Segments for installation in a specific aircraft type, the airline's aircraft engineering personnel meet with LiveTV engineers to jointly develop an installation drawing package. A tray location is identified and selected from available options in the avionics equipment bay. The best way to get power to the tray is determined and a circuit breaker is assigned in the flight deck. Interfaces to other avionics equipment and aircraft discretes are similarly identified and installed. All new wiring must comply with strict federal regulations and standards for installation.

A location for the aircraft antenna is chosen on the top side of the aircraft taking into account availability, accessibility, electromagnetic compatibility, and aerodynamic efficiency. Once defined, new drawings are created that define how the aircraft skin is penetrated and a doubler plate installed for antenna mounting, structural support, and environmental sealing. A nearby location for the LNA/PA is chosen between the passenger cabin and the aircraft skin. The closest stanchion is identified and the method of mounting the LNA/PA is documented. The RF Cable is then pulled through the aircraft to its destination. The cable is routed and restrained in compliance with specified installation standards. The cable is cut to length and terminated at both ends. The power level at the input to the RF Assembly is checked to make sure that it is within the input dynamic range of the amplifier. All details of the modification are carefully documented and inspected by the FAA.

The installation is verified by using an installation test set to verify functionality. A variety of ground tests are performed to assess the electromagnetic compatibility of the newly installed equipment with existing flight critical equipment. An STC Aircraft Test is then performed to verify the electromagnetic compatibility of the newly installed equipment with existing flight critical equipment. At the conclusion, an STC package consisting of the drawing package and test data is prepared. The STC package is then submitted for review by the FAA. Upon their approval, an STC for the defined type of aircraft is granted. Once granted, the STC is only valid for that specific aircraft type. If the airline desires to install the WADL Aircraft Segment in another type of aircraft, the process is repeated. All aircraft of a specific type are modified in exactly the same fashion based on the STC drawing package. Installations are carefully inspected and discrepancies are documented and properly corrected.

The installation and controls process are such that the system will always be configured as per the installation design. The FAA certification requires that all cables and system pieces be in place per installation drawing, else the aircraft is not allowed to fly. The air transport industry carefully controls the configuration of installed avionics equipment. The FAA issues a Supplemental Type Certificate for the aircraft on the basis of the newly installed, tested, and approved equipment. Any changes to the installation or hardware baseline would require their approval. LiveTV has and will continue to state that tampering with or modifying installed equipment could result in damage to the installed equipment and a violation of FCC regulations.

<u>Conclusion:</u> Taken as a whole the STC process, professional installation per documented Work Orders, and strict FAA regulations with respect to aircraft maintenance and configuration will insure that the system as installed and maintained will remain compliant as defined in FCC Part 15C and verified by the TCB.

### **Attachment A**

### **Example Installation Work Orders**



#### LiveTV WORK ORDER and IN PROGRESS INSPECTION FORM REPAIR STATION NO. L1IR073X

	NET AIR OTATION NO. ETIROTOX		
PART No.o	r A/C TAIL No: <u>N906FR</u> S/N: <u>1684</u> TYPE A/C:	•	319
CUSTOME	R: Frontier W.O. NO.: F90300303 DATA:	MDL 99FS9	
	Document Number: LTV933302, Sh. 104, Rev. A.		
	Document Number: LTV3092090 Rev. A.		
	Document Number: LTV3092113 Rev. N/C.		
ITEM	NARRATIVE "WADL System Elec. Install., Pg. 1 - 2."	MECH	DATE
	NOTE: This equipment must be installed and operated IAW provided instructions and a minimum 20cm spacing must be provided between computer mounted antenna and a persons body (excluding extremities of hands, wrist and feet) during wireless modes of operation. Optional vehicle mounted antenna must not exceed 6.0 dBi antenna gain and should be professionally installed. This device complies with Part 15 of the FCC Rules. Operation is subject to the complies with Part 15 of the FCC Rules. Operation is subject to the following two conditions: (1) this device may not cause harmful interference, and (2) this device must accept any interference received, including interference that may cause undesired operation. CAUTION: Any changes or modifications not expressly approved by the party responsible for compliance could void the authority to operate equipment.		
1	Check to ensure A/C power is off. If necessary, remove the power from the aircraft. Gain access to the 9001VT Terminal Block.		
2	IAW dwg LTV3092113 install WADL System wire support structure at C63 stringer 20R.		
3	IAW dwg LTV3092090 route all system electrical cables.		
4	IAW dwg LTV933302 terminate WADL cable ends.		
5	IAW dwg LTV3092090 install required circuit breakers for the WADL system.		
6	IAW dwgs LTV3092090 and LTV933302 terminate stowed wiring IAW Airbus standard wiring practices and termination sheets.		
7	IAW dwg LTV933302 coil and stow wiring at the 80VU rack area.		
8	Install and route wiring harness between 9001VT to PFS001 disconnect area. Coil and stow at location area 9000VT PFS001 termination area.		
9	IAW dwgs LTV3092090 and LTV933302 install and route coax harness between the WADL antenna and the LNA.		
10	IAW dwg LTV933302, install and route the coax harness between the LNA and the PFS001 rack disconnect.		
		Q.C. Inspector	<del></del>

SUPERVISOR:\_\_\_ DATE: Stamp (print) (sign)



REPAIR STATION NO. L1IR073X

PART No.c	or A/C TAIL No:	A-3	319
	CUSTOMER: Frontier Airlines W.O. NO.: F90228302 DATA		
ITEM	NARRATIVE "wadl system,3.1, elec installation"	MECH	DATE
I I ⊏IVI	Check there is no power on the aircraft, if necessary, remove the	MECH	DATE
1	power from the aircraft.  Gain access to the 9001VT Terminal Block.		
2	Per Harris Dwg 3092113 install WADL system wire support structure at C63 stringer 20R		
3	Per Harris Dwg 3092090 route in all system electrical cables.		
4	Per Harris Dwg 3092095 terminate WADL cable ends.		
5	Per Harris Dwg 3092090 Install required circuit breakers for the WADL systems.		
6	IAW Dwgs 3092090 and 3092095 terminate stowed wiring IAW Airbus standard wiring practices and termination sheets.		
7	IAW Deg 3092095 Coil and Stow wiring at the 80VU rack area.		
8	Install and route wiring harness between 9001VT to PFS001 disconnect area, coil and stow at location area 9000VT PFS001 termination area.		
9	IAW Dwgs 3092090 and 3092095 and PAD 2284-920 and -923 install and route coax harness between the WADL antenna and the LNA		
10	IAW Dwg 2284-920 and -923 Install and route the Coax harness between the LNA and the PFS001 rack disconnect.		
SUPERVISOR	: SIGN: DATE:	Q.C. Inspector Stamp	



REPAIR STATION NO.L1IR073X

		411 017 11011 110	J.L 111 (07 07)			
PART No.	or A/C TAIL No:	S/N:_		TYPE A/C:	A3	19
	CUSTOMER: Frontier Airlin	<u>es</u> W.O. NO.:	F90228302	DATA:		
ITEM	NARRATIVE "WAD	L System ta	lly sheet"		MECH	DATE
1	NOT USED				х	х
2	Verify accomplishment WA installation, Items 1 throug		/ork order; 2, Me	ech		
3	Verify accomplishment WA Instalation, Items 1 through		/ork order; 3.1,	Elec		
4	Verify accomplishment WA Items 1 through 8	ADL System W	/ork order; 4, Clo	ose Up,		
5	Verify accomplishment WA Items 1 through 9	ADL System W	/ork order; 5 Act	ivation		
SUPERVISOF	R:SIGI	N:(sigi	DATE:		Q.C. Inspector Stamp	



REPAIR STATION NO. L1IR073X

PART No.o	or A/C TAIL No: S/N:	TYPE A/C:	A-3	319
	CUSTOMER: Frontier Airlines W.O. NO.: F90	228302 DATA:	·	
ITEM	NARRATIVE "wadl system,2,mech	installation"	MECH	DATE
1	Remove cargo liner sidewalls to gain access to Note: N/A this step and steps 2, 3, and 4 if rack is installed.			
2	per harris dwg 3092107, install the wadl rack s accordance with Airbus SRM chapter 51 airfra practices			
3	per harris dwg 3092089 install the wadl structu	ire components		
4	per harris dwg 3092091 install the wadl systen	n components		
5	Gain access to the aircraft ceiling center line be overhead ceiling panels between C60 and C6	•		
6	per Harris dwg 3092110 install the antenna structu C61 and stringers 1L and 1R. Ensure all workman with Airbus SRM chapter 51 airframe standard pra	ship is in accordance		
7	per harris dwg 3092110 install the wadl antenr	na.		
8	Inspection: Inspect wadl antenna installation a	nd doublers		<b>&gt;</b>
9	per Harris dwg 3092112 install wadl LNA structure and C60, stringers 4R/5R. In accordance with airframe standard practices			
10	Inspection: Inspect wadl LNA support structure	e installation.		
SUPERVISOR	: SIGN: (print) (sign)	DATE:	Q.C. Inspector Stamp	$\langle \rangle$



REPAIR STATION NO. L1IR073X

PART No.	or A/C TAIL No: TYPE A/C:	A-3	319
	CUSTOMER: Frontier Airlines W.O. NO.: F90228302 DATA:		
ITEM	NARRATIVE "wadl system,5,activation"	MECH	DATE
1	Gain access as required to perform the following tests		
2	Not Used	Х	х
3	per Harris dwg 3092090 remove collars and close all WADL circuit breakers.		
4	per Harris dwg 3092082 Apply power to the aircraft and perform the wadl system functional test.		
5	Ensure all tools and any foreign material are removed from all areas that were accessed for the wadl system		
6	Inspection: Ck areas for insulation blankets and foreign materials. Give OK to close when inspection completed.	$\Diamond$	
7	Close all access panels opened in item 1.		
8	make the following logbook entry to cover the WADL activation: "THE WIRELESS DATA LINK HAS BEEN ACTIVATED IN ACCORDANCE WITH PART 5 OF EO 2302095."		
9	Prepare the aircraft for return to service.		

SUPERVISOR: SIGN: DATE: Q.C. Inspector Stamp



REPAIR STATION NO. L1IR073X

	RELAIR STATION NO. ETIROTOX		
PART No.o	r A/C TAIL No:	A-3	319
	CUSTOMER: Frontier Airlines W.O. NO.: F90228302 DATA:		
ITEM	NARRATIVE "wadl system,3.2, elec installation"	MECH	DATE
11	IAW dwg 2284-920 and -923 terminate Coax cable. Terminate all wiring in accordance with Airbus Standard wiring Practices.		
12	Ensure all wiring for WADL and CSS circuit breaker provision is installed IAW Dwgs 3092090 and 3092095 and PAD drwgs 2284-920 and 2284-923		
	NOT USED		
SUPERVISOR	SIGN: DATE: (print) (sign)	Q.C. Inspector Stamp	



REPAIR STATION NO. L1IR073X

PART No.	or A/C TAIL No:	S/N:		TYPE A/C:	A-3	319
	CUSTOMER: Frontier	Aiorlines W.O. NO.:	F90228302	_ DATA:		
ITEM	NARRATIVE	"wadl system,4 c	lose up"		MECH	DATE
1	Perform continuity ch disturbed during the		aircraft wires th	nat were		
2	Ensure that all tools a work areas that were					
3	In the aft cargo work closing the area for; cattachment studs, an	clean & dry, not dar	naged, Velco ta	•		
4	Inspection: Okay to c	lose work area pane	els		$\Diamond$	
5	Close all panels remo	oved for the WADL s	system installat	ion		
6	Make the following lo MAJOR alteration of "THIS AIRCRAFT HAINSTALLING THE W	this aircraft: AS BEEN MODIFIE	D BY E.O. 230	2095		
7	Complete the attache	ed weight change no	tification form.			
8	Prepare the aircraft for	or service				
SUPERVISOR	(print)	SIGN: (sign	DATE:		Q.C. Inspector Stamp	

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