

Figure 3-54 Radiated Emissions, Narrow Band 9.0 to 12.0 GHz

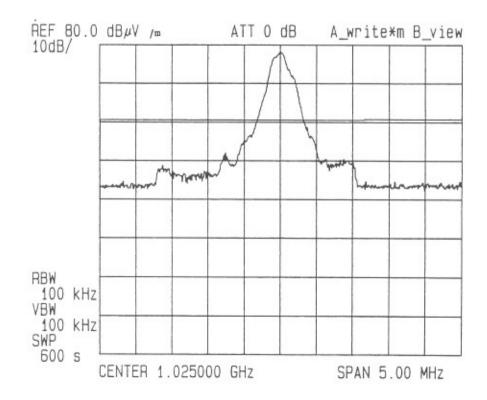


Figure 3-55 Radiated Emissions, Narrow Band Centered on Channel 1X

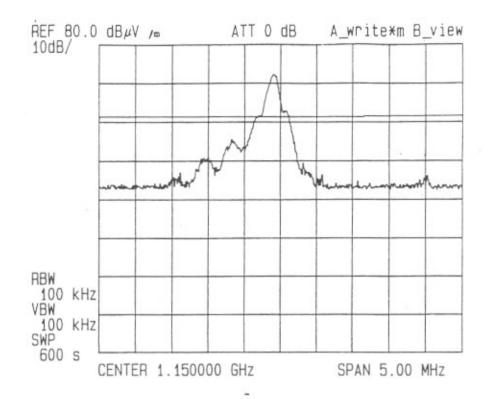


Figure 3-56 Radiated Emissions, Narrow Band Centered on Channel 126X

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### 3.6 Frequency Stability (2.995)

The test equipment was set up as shown in Figure 3-57. The spectrum analyzer was set to a slow sweep and into a storage mode. The peak of the spectrum was assumed to be the carrier frequency and the maximum deviation of the peak from the desired carrier frequency was recorded as the error. An example of this technique is shown in Figure 3-58 for channel 1X and Figure 3-59 for channel 64Y.

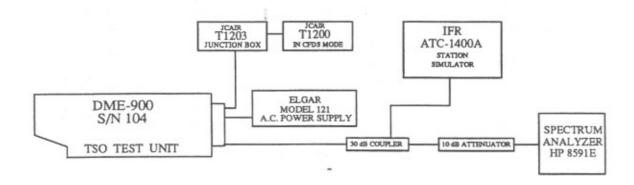


Figure 3-57 Equipment Set-up For Measuring Frequency Stability At The Antenna Port

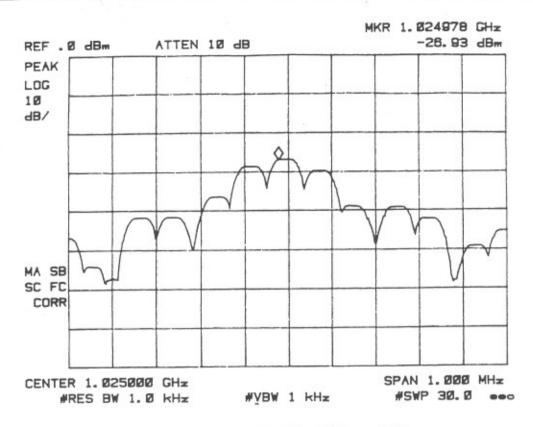


Figure 3-58 Frequency Stability of Channel 1X

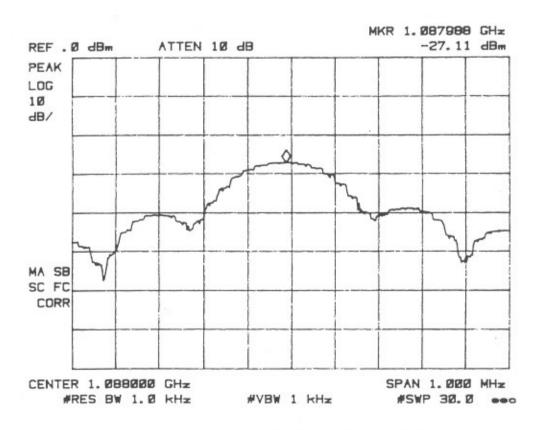


Figure 3-59 Frequency Stability of Channel 64Y

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#### 3.6.1 Frequency Stability With Temperature Variation (2.995 (a, b) (2))

The DME-900 was placed in the temperature chamber and stabilized on temperatures of -20,- 10,0,10,20,30,40, and 50 deg c. The Local Oscillator (LO) output frequency (L-Band) was measured at the antenna port are recorded for each temperature for both X and Y channels.

## 3.6.1.1 Frequency Vs Temperature X-Channels

		Frequency Error + *						
CHAN	-20 C	-10 C	0 C	+10 C	+20 C	+30 C	+40 C	+50 C
1	+50	+45_	+33	+35_	+28	+23	25_	-25
12	+48	+38	+33	+35	+25	30_	30	30_
23	+43	+33	+28	+25	-35	30_	-32	-32
34	+50	+40	+35	+35_	+30	+28	-25	-22
45	+55	+45	+40	+35_	+35	+25	-27	-22
56	+58	+40	+40	_+40_	+35	+28	-30	-22
67	+55	+45	+40	_+40_	+28	+28	-25	-20
78	+48	+43	+43	_+40_	_+30_	22_	-30	27_
89	+45	+38	+33	+28	_+20_	-27	-30	30_
95	+43	+30	+33	+25	+20	-30	-35	-25
100	+40	+35	+28	+28	+23	-30	30	-30
111	+33	+33	+20	+20	-32	-25	-32	-32
126	+35	+28	+25	+28	+18	-30	-32	27_

<sup>\*</sup> data shown is the error in KHz from the assigned channel frequency

#### 3.6.1.2 Frequency Vs Temperature Y-Channels

			Frequen	cy Error +	k			
CHAN	-20 C	-10 C	0 C	+10 C	+20 C	+30 C	+40 C	+50 C
1	-10	+08	+05	-07	-10	_0_	-02	-05
12	-07	-10	-07	0	-02	-02	-30	-22
23	-30	-25	-17	-17	-10	-25	-30	-37
34	-10	-07	-15	-17	-17	+03	-07	-10
45	-10	+03	-12	-02	-17	-20	-20	-05
56	-20	-12	+03	-17	-07	0	-20	-10
67	-30	-10	-12	-02	-02	-20	-12	-10
78	-12	-12	-02	+10	-17	0	-10	-20
89	-17	+03	-20	-27	-07	-25	-20	-20
95	+10	-10	20	-02	-10	-17	-22	-25
100	+13	-12	-15	0_	-17	-27	-32	-07
111	+05	-27	-10	-25	27_	-32	-32	-35
126	22_	+03	-12	+08	-02	-05	-17	-07

<sup>\*</sup> data shown is the error in KHz from the assigned channel frequency

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## 3.6.2 Frequency Stability With Voltage Variation (2.995 (d) (1))

The LO output was measured at the antenna port with the line voltage set at 85%, 100%, and 115% of the nominal line voltage of 115 vac.

## 3.6.2.2.1 Frequency Vs Line Voltage X-Channels

				FREQ ERROR	
CHAN	FREQUENCY	97.75(85%)		115(100%)	132.25(115%)
1	1025	-22		25	30
12	1036	-27		-32	-35
23	1047	-32		42	_40_
34	1058	-27		30	35
45	1069	-30		35	37
56	1080	-27		-35	-40
67	1091	-30		-32	-37
78	1102	-27		30	35
89	1113	-25		-22	-27
95	1119	-30		35	37
100	1124	-27		-35	37
111	1136	-27	-	-35	35
126	1150	-37		42	-40

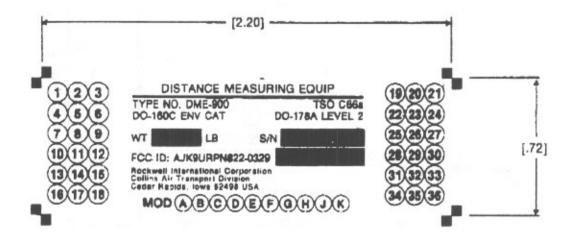
<sup>\*</sup> data shown is the error in KHz from the assigned channel frequency

#### 3.6.2.1 Frequency Vs Line Voltage Y-Channels

	ASSIGNED		FREQ ERROR	*
CHAN	FREQUENCY	97.75(85%)	115(100%)	132.25(115%)
1	1025	-20_	10	12
12	1036	-15	10	07
23	1047	-05	12	-15
34	1058	-22	20	17
45	1069	-10	02	07
56	1080	07	17	17
67	1091	40	40	22
78	1102	_40	-40	40
89	1113	30	-35	37
95	1119	-10	-12	15
100	1124	-10	07	07
111	1136	0	+05	+23
126	1150	_42	-42	_40_

<sup>\*</sup> data shown is the error in KHz from the assigned channel frequency

# 4. EQUIPMENT IDENTIFICATION PLATE (2.993 (f))



685-7297-001 B

Figure 4-1 DME-900 Equipment Identification Plate

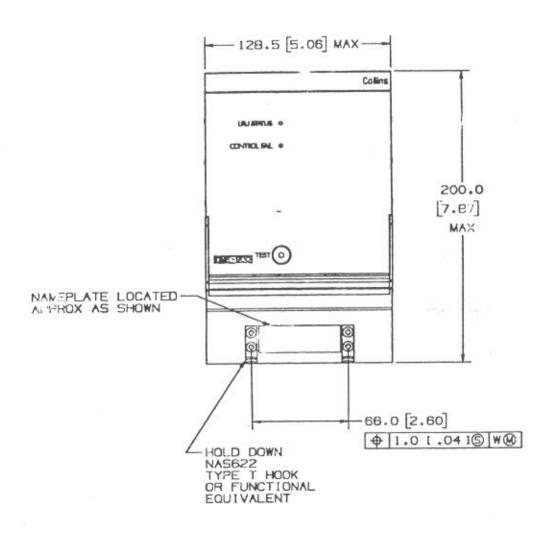


Figure 4-2 DME-900 Equipment Installation Drawing

# 5. EQUIPMENT PHOTOGRAPHS (2.993 (g))

## 5.1 List Of Photos

Figure 5-1	DME-900 Total Unit
Figure 5-2	DME-900 Right Side Door
Figure 5-3	DME-900 Right Side Door Showing SMO and Driver
Figure 5-4	DME-900 Right Side Chassis
Figure 5-5	DME-900 Right Side Chassis Showing PA and Receiver
Figure 5-6	DME-900 Left Side Door Showing Digital Card
Figure 5-7	DME-900 Left Side Chassis Showing Power Supply

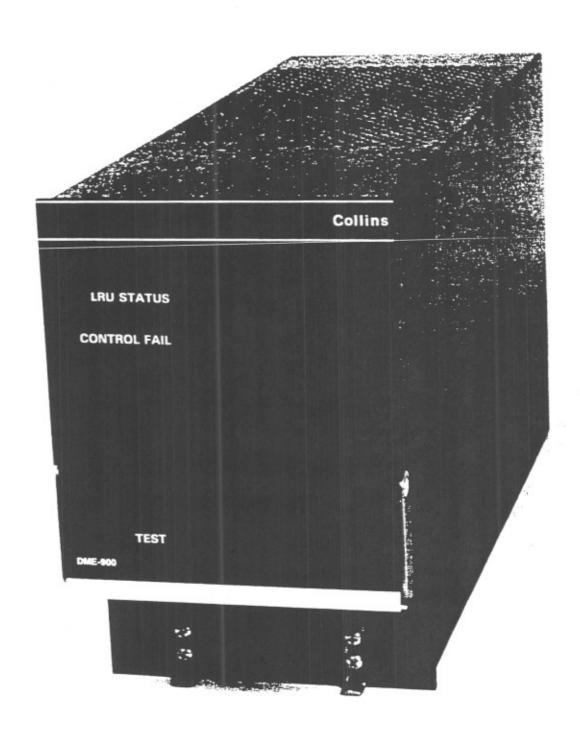


Figure 5-1 DME-900 Total Unit

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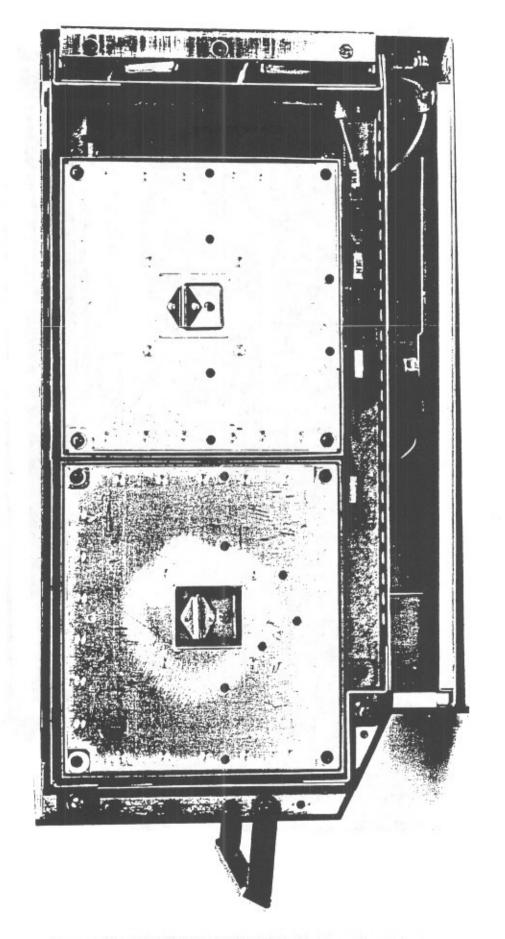


Figure 5-2 DME-900 Right Side Door

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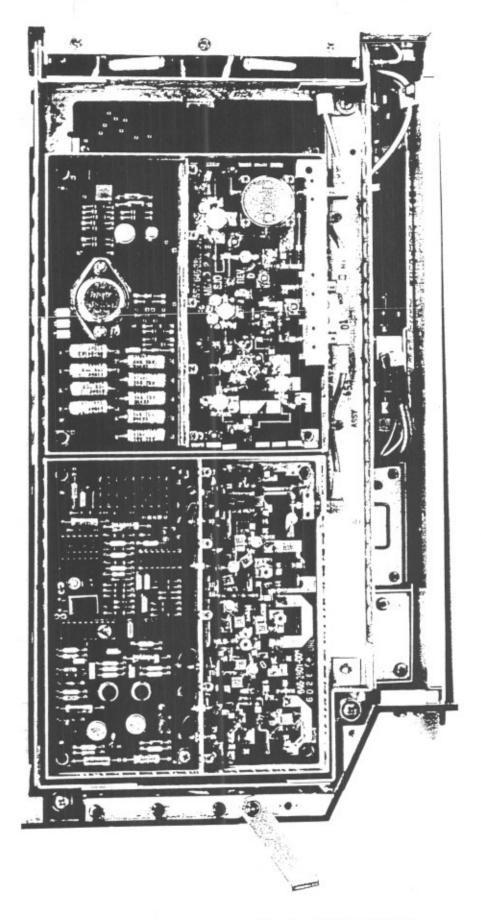


Figure 5-3 DME-900 Right Side Door Showing SMO and Driver

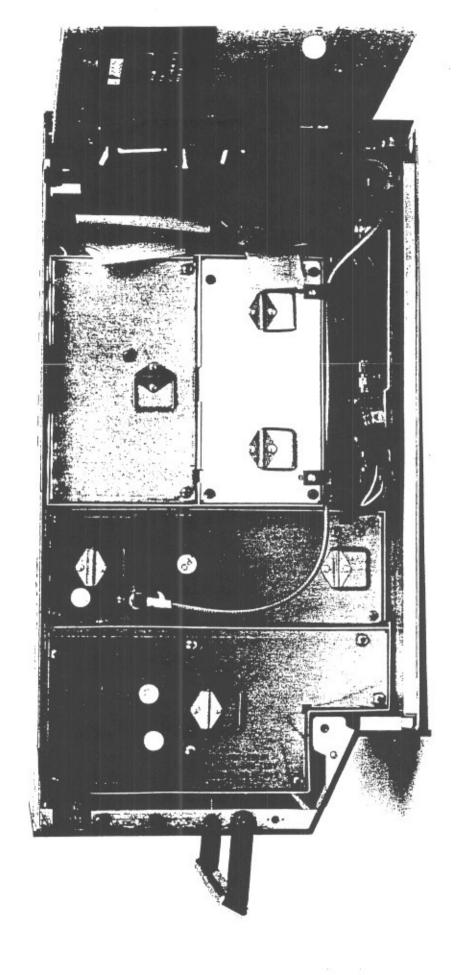


Figure 5-4 DME-900 Right Side Chassis

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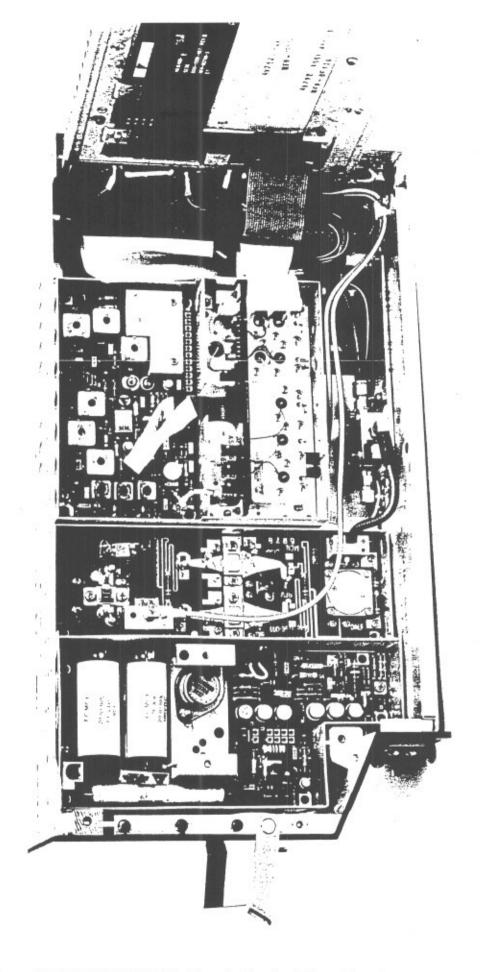


Figure 5-5 DME-900 Right Side Chassis Showing PA and Receiver

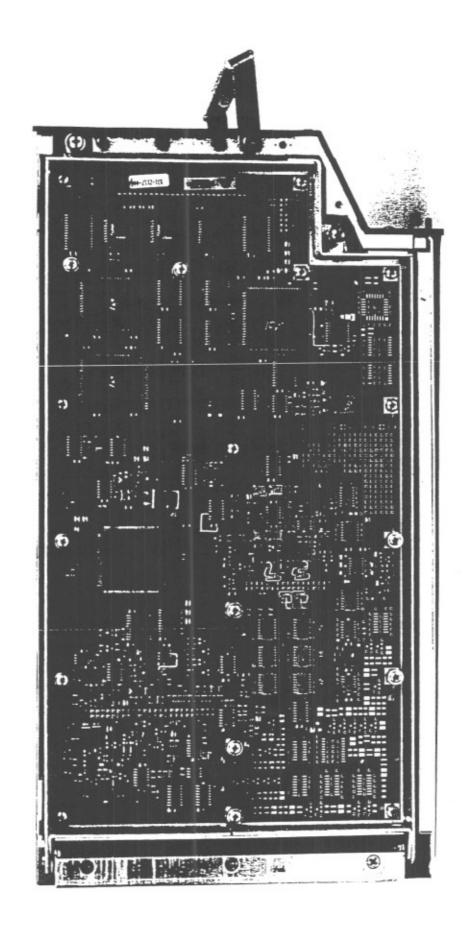


Figure 5-6 DME-900 Left Side Door Showing Digital Card

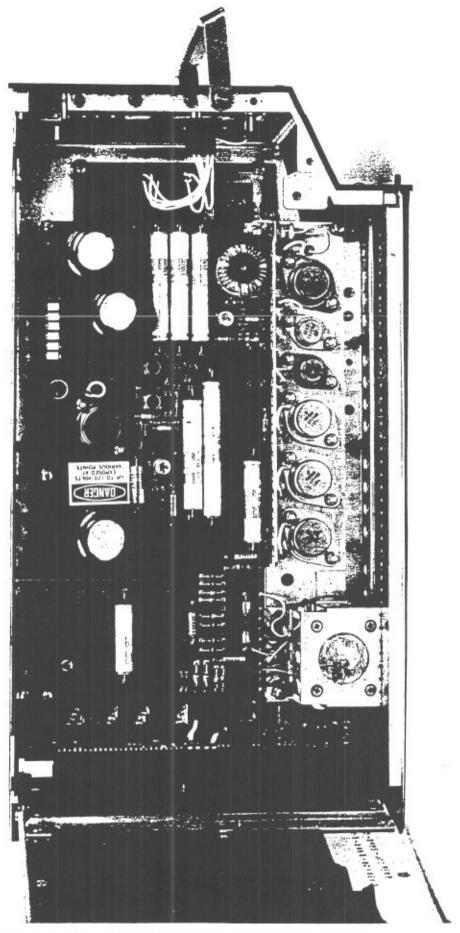


Figure 5-7 DME-900 Left Side Chassis Showing Power Supply

## 6. QUALIFICATIONS

#### 6.1 Test Engineer

Gregory S. Haywood Design Engineer Pulse Navigation Section Air Transport Engineering

#### Present Assignment

Hardware Design Engineer for the DME-900.

#### Experience at Rockwell Collins

1991 - 1994 Design Engineer DME-900 Collateral Design Engineer, DME-700 and 860E-5

1989 - 1991 Reliability Engineer, Commercial Avionics Products (ILS-720, TDR-94D, AHC-86, DGS-65, DAU-850,SDD-640)

1987 - 1987 Component Application Engineer

## 6.2 Oath Of Attestation, Project Engineer

The technical data supplied with this application has been taken under my supervision and is hereby duly certified.

Signed

R.J Oswalt

Project Engineer

DME-900

#### Experience at Rockwell Collins

1991 - 1994Project Engineer for the DME-900

1990 - 1991 Design Engineer, DME-700 ALT-55

1988 - 1990 Design Engineer TDR-94

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eer 29 July '94

# 6.3 Oath Of Attestation, Supervisor

I certify that the above application was prepared under my direction and to the best of my knowledge and belief,the facts set forth in the application and accompanying technical data are true and correct.

Signed

C.E. Steen

Manager, Pulse Navigation Section

Air Transport Engineering